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FY 79 - DEVELOPMENT OF FIBER OPTICS  
CONNECTOR TECHNOLOGY FOR LARGE SPACE SYSTEMS

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DEVELOPMENT OF FIBER OPTICS  
TECHNOLOGY FOR LARGE SPACE STRUCTURES

OBJECTIVE:

TO DEVELOP PHYSICAL CONCEPTS FOR INTEGRATING FIBER OPTIC CONNECTORS AND CABLES WITH STRUCTURAL CONCEPTS PROPOSED FOR LSST. EMPHASIS IS PLACED ON REMOTE CONNECTIONS USING INTEGRATED CABLES.

The inherent advantages in using a fiber optics communications system are listed below

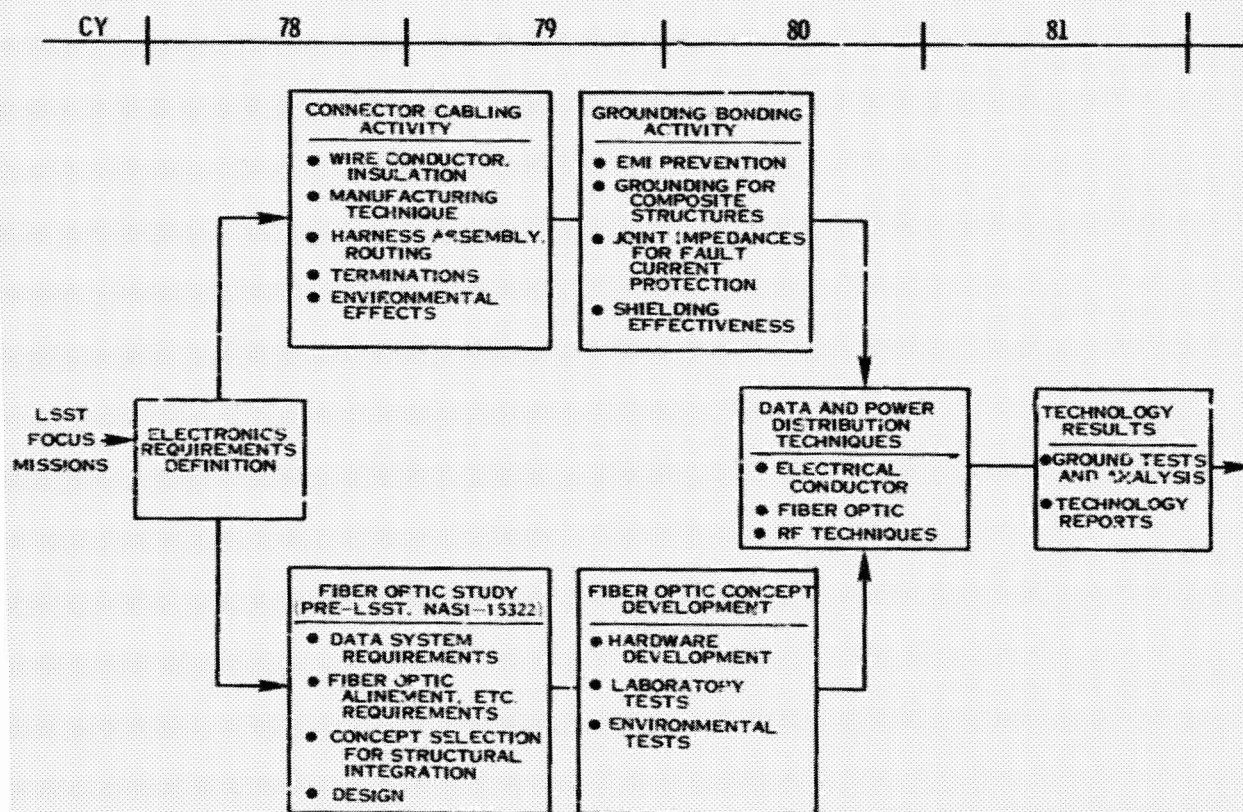
#### F/O COMMUNICATION SYSTEM QUALITIES

- ° HIGH DATA RATES
- ° NO CROSS TALK
- ° NO EMI
- ° SECURE COMMUNICATIONS
- ° LIGHT WEIGHT



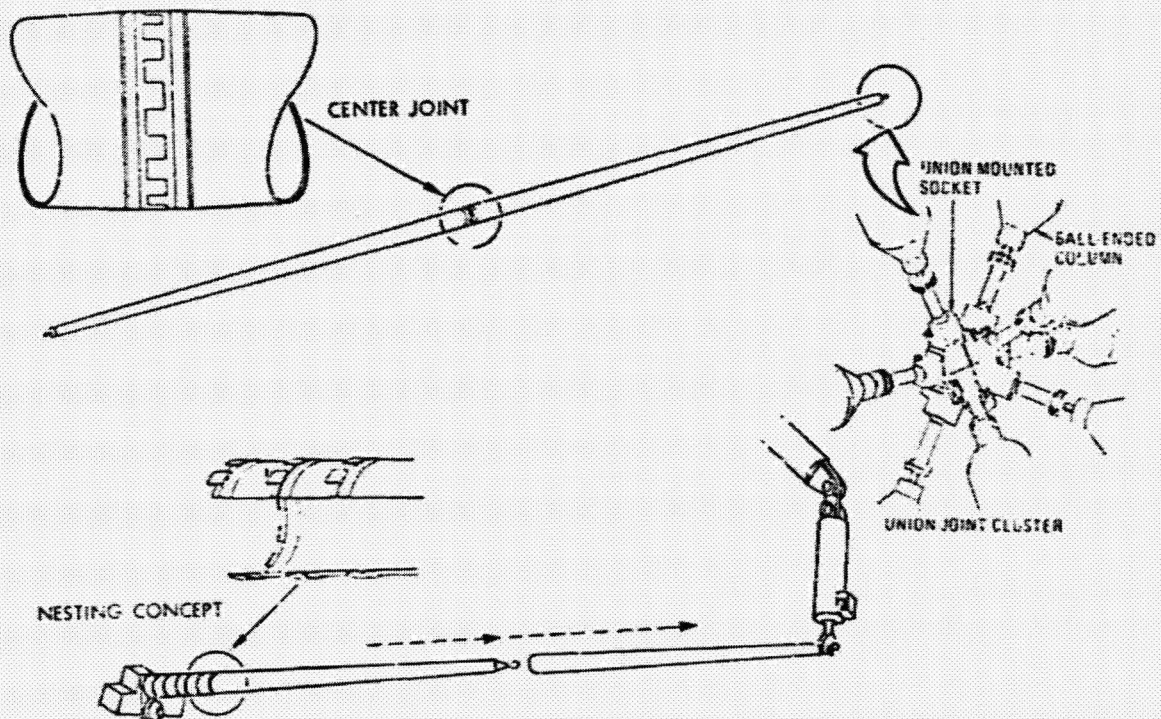
The LSST Electronics development plan is shown below. This plan separates the fiber optic activity from the basic connector/cabling activity that is under the management of the Marshall Space Flight Center.

## LSST ELECTRONICS DEVELOPMENT PLAN



The fiber optics development effort for large space structures was initiated with a specific structured element in mind, specifically, the nestable column as shown below. The F/O connector is being developed to be compatible with the union joint cluster and the assembly concept using the End-Effector of the Shuttle RMS.

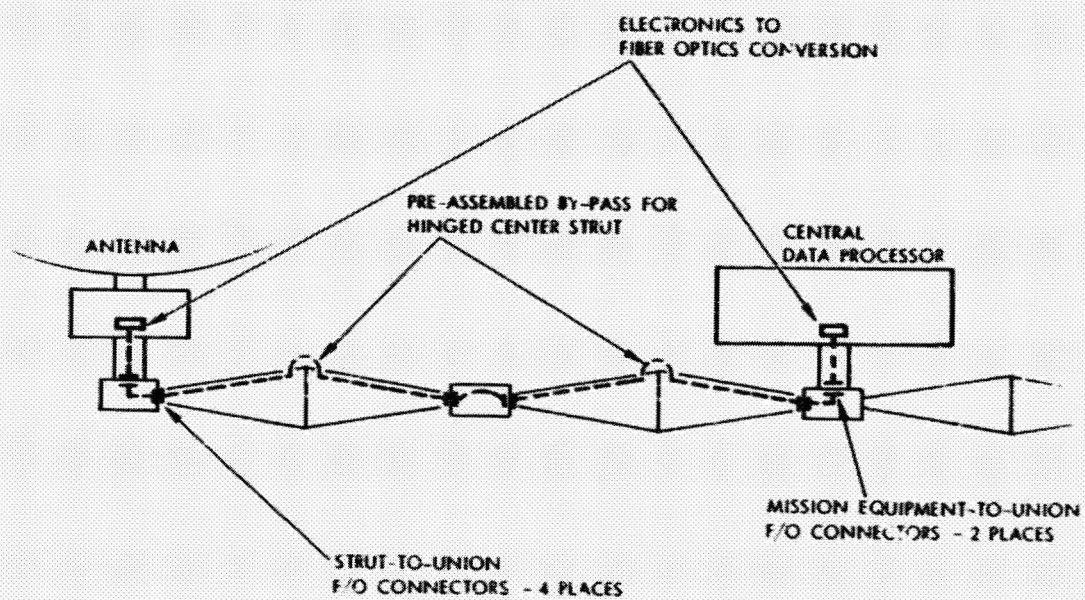
#### NESTABLE COLUMN





This figure shows the basic concept of integrating the F/O connector into the large space structure.

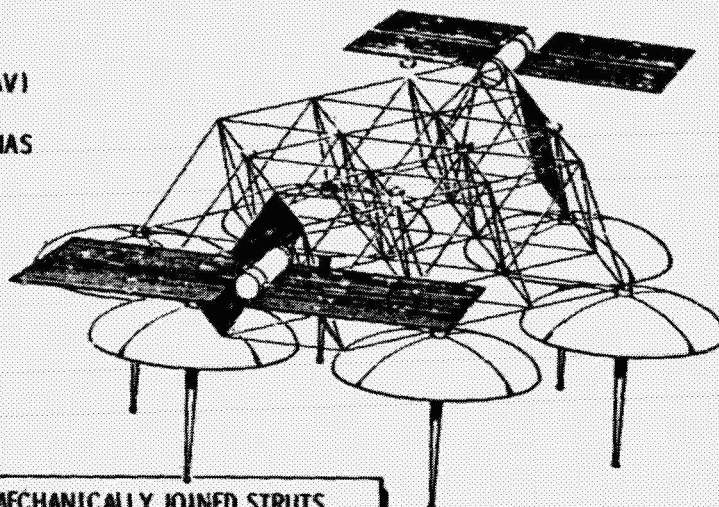
#### ANTENNA TO CENTRAL PROCESSOR CABLE PATH



The Electronic Mail Satellite (EMS) concept shown below was used to define the requirements for the fiber optics development effort.

**TASK 1 MODEL SYSTEM  
ELECTRONIC MAIL SATELLITE**

550 POSTAL CENTERS  
48 M PAGES MAIL/24 HRS (AV)  
5 OPERATING AREAS  
5 ACTIVE IN-ORBIT ANTENNAS  
+1 SPARE  
110 CHANNELS/ANTENNAE



STRUCTURE: MECHANICALLY JOINED STRUTS  
129 STRUTS, EA 15M LONG  
33 UNIONS, EA 17 CM DIA

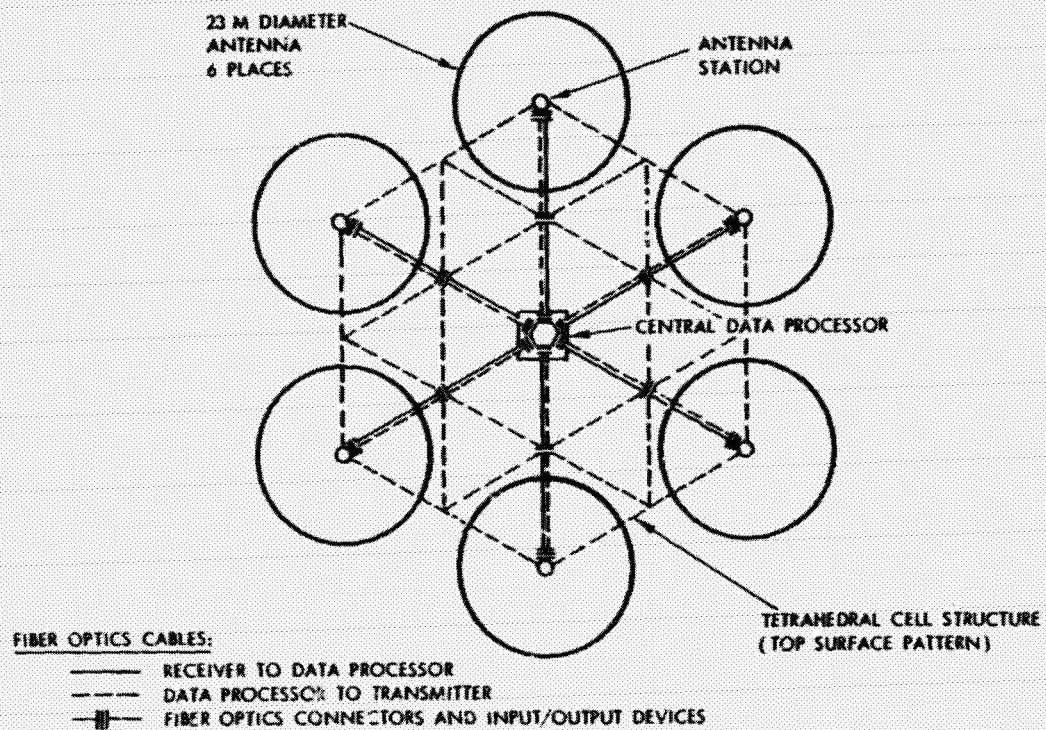
DIMENSIONS: LENGTH 96M  
WIDTH 61M  
HEIGHT 12M

WEIGHT: 37,280 LB (DRY)



A schematic wiring diagram for the conceptual EMS system is shown below.

# SCHEMATIC WIRING DIAGRAM, COMMUNICATIONS, FOR EMS MODEL



The requirements proposed for the Electronic Mail Satellite mission scenario are presented below. Also presented is a comparison of various cable systems and the weight reduction that a typical F/O system would have over conventional systems.

#### REQUIREMENTS PROPOSED FOR ELECTRONIC MAIL SATELLITE

- ° FREQUENCY: 3 GHz
- ° 110 CHANNELS PER STATION (6 STATIONS)
- ° 4 DATA LINES PER STATION
- ° 4 CONTROL LINES PER STATION
- ° SEGMENT LENGTH: 32.2 METERS
- ° FIVE REFLECTORS, ONE SPARE
- ° TOTAL CABLE REQUIRED: 42.5 km (141,666 FEET)
- ° DATA RATE = 85 MBPS PER DATA LINE
- ° BANDWIDTH = 100 MHz
- ° BER =  $10^{-9}$
- ° S/N = 13 dB
- ° F/O LINES PER BRANCH: 8

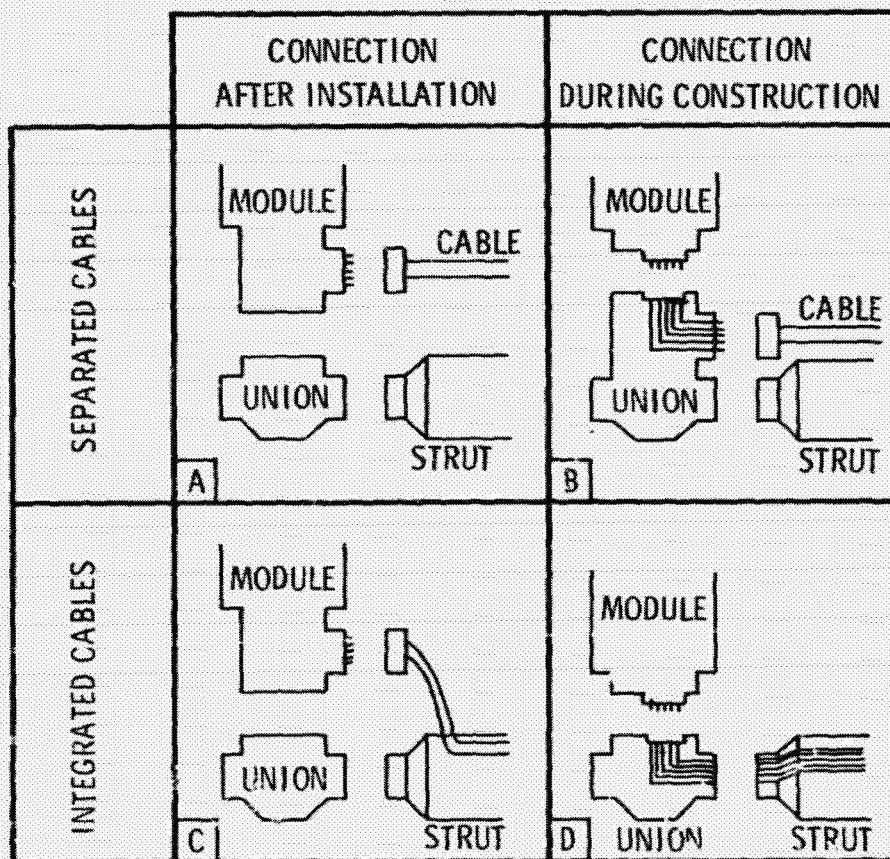
#### CABLE COMPARISON

CABLE	LOSS	WEIGHT
SHIELDED TWISTED PAIR	198 dB/km	4,014 LB
SEMI-RIGID COAX	92 dB/km	11,687 LB
FIBER OPTIC	294	944 LB
	20	41 LB



The various power and signal installation options are shown below. Option D is the desired configuration for the F/O distribution system.

## POWER AND SIGNAL INSTALLATION OPTIONS



Based on requirements for the EMS system and an assessment of F/O technology, the following critical technologies were identified.

### CRITICAL TECHNOLOGY IDENTIFIED

#### 1. F/O CABLE TECHNOLOGY

- ° A SINGLE FIBER AND/OR MULTI-FIBER BUNDLES ARE REQUIRED THAT CAN WITHSTAND RADIATION EFFECTS ( $10^7$  RADS) AND TEMPERATURE EXTREMES AS WELL AS PROVIDE SUFFICIENT CORE DIAMETER FOR CONNECTOR ALINEMENT.

(NO CABLE EXISTS AT THIS TIME THAT CAN MEET THESE REQUIREMENTS.)

#### 2. CONNECTOR TECHNOLOGY

- ° A CONNECTOR IS REQUIRED THAT WOULD BE COMPATIBLE WITH CABLE DESCRIBED ABOVE.
- ° AN ASSEMBLY CONCEPT IS REQUIRED FOR MATING F/O CONNECTORS REMOTELY WITHIN STATE-OF-THE-ART LOSSES.

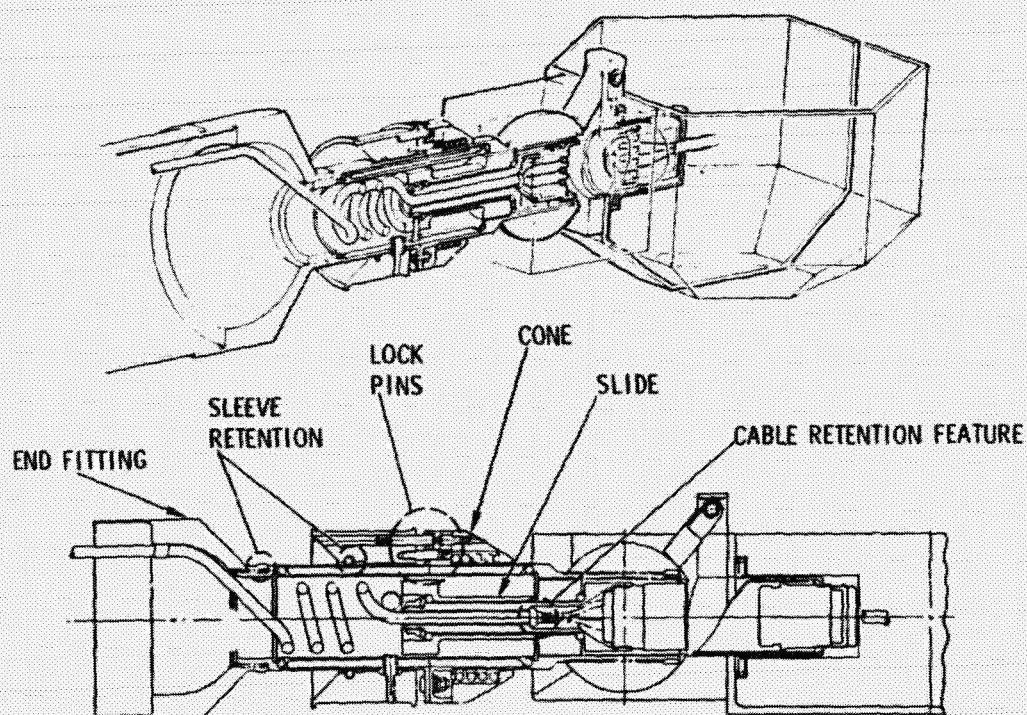
#### 3. STRUCTURE INTEGRATION TECHNOLOGY

- ° METHODS MUST BE DEVELOPED FOR INTEGRATING F/O COMPONENTS INTO STRUCTURAL ELEMENTS.



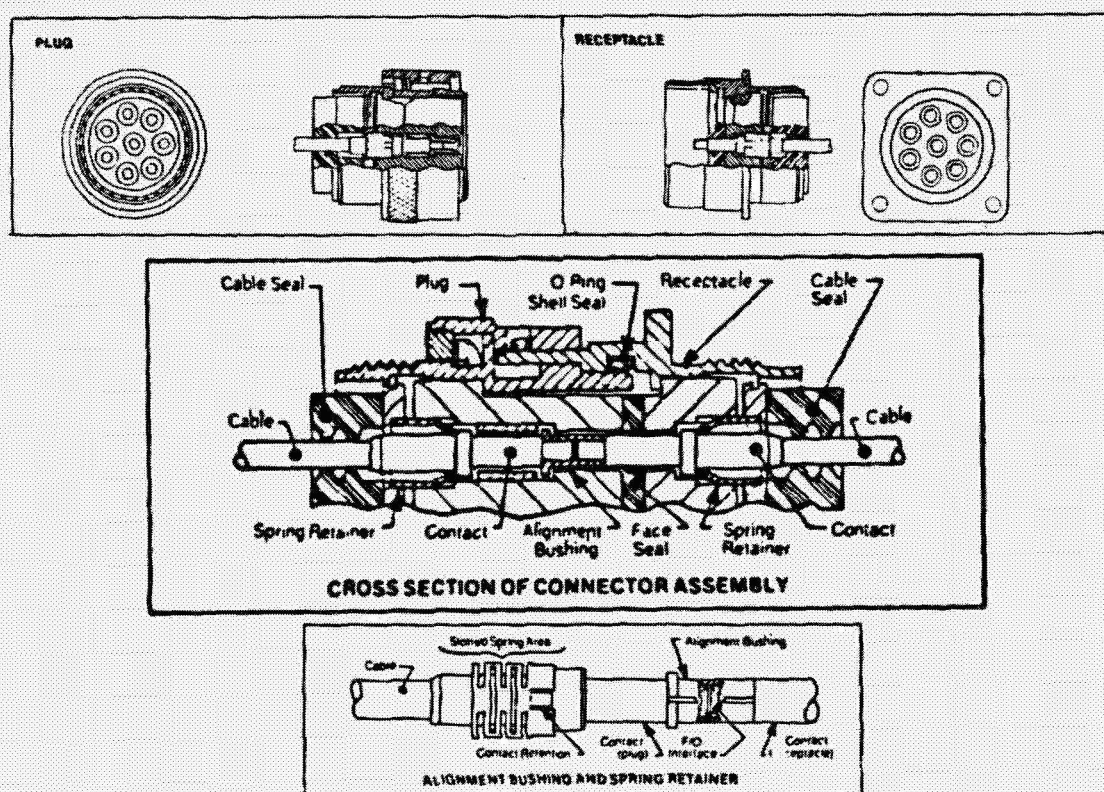
The F/O integrated connector assembly presently being developed by Rockwell International is shown below. This connector assembly shall be used to test two commercially available F/O connectors - (Amphenol and Hughes). These connectors will be assembled and tested so that the various system losses can be determined.

#### INTEGRATED CONNECTOR ASSEMBLY



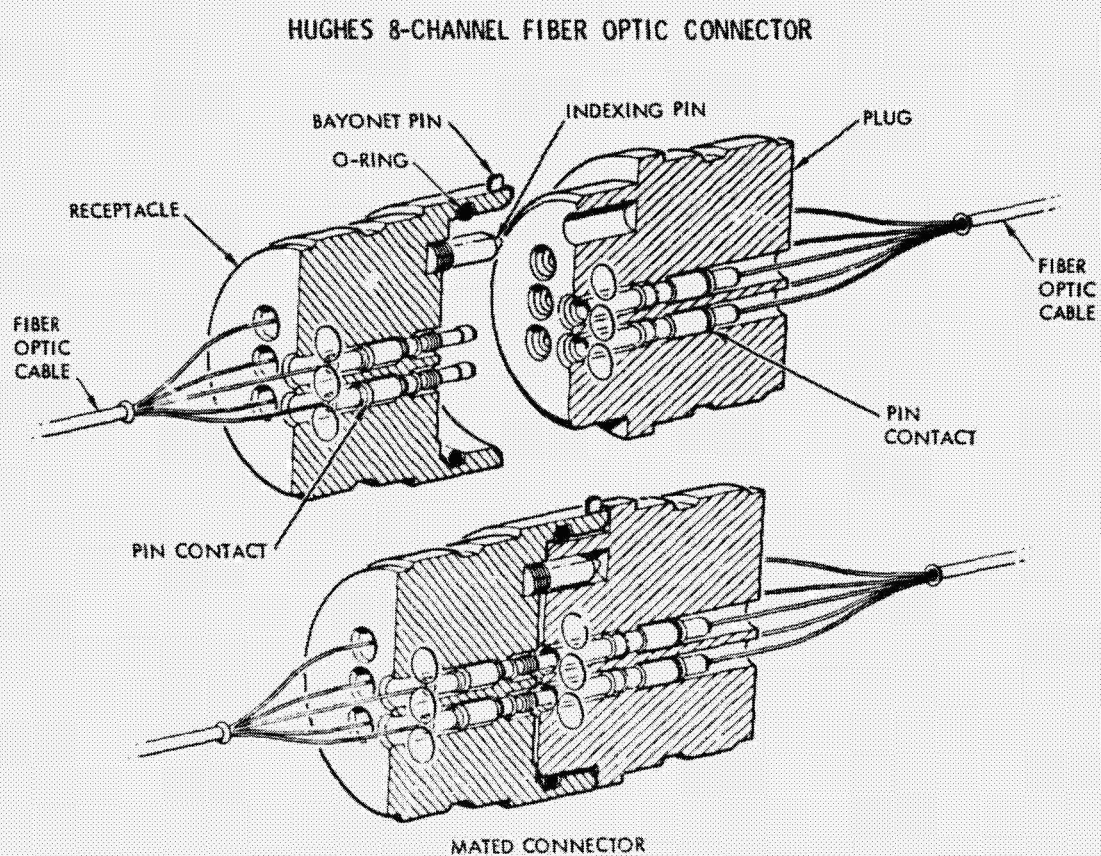
The Amphenol 8-Channel F/O connector is shown below.

### AMPHENOL 8-CHANNEL FIBER OPTIC CONNECTOR





The Hughes 8-Channel F/O connector is shown below.



After the F/O Integrated Connector Assembly is fabricated, the following tests shall be conducted.

## LARGE SPACE STRUCTURES

### PROPOSED CONNECTION TESTS

- ° MEASURE FIBER OPTICS LIGHT LOSSES OF ASSEMBLY
- ° PERFORM REPEATED MATE/DEMATE OPERATIONS
- ° MEASURE LIGHT LOSSES DURING TEST CYCLE
- ° IDENTIFY LOSS PORTIONS
- ° MODIFY CONNECTION ASSEMBLY TO REDUCE LIGHT LOSSES
- ° REPEAT CYCLIC TESTS
- ° PREPARE TEST REPORT